

Causal Relationships between Solar Proton Events and Single Event Upsets for Communication Satellites

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In this work, we analyze a historical archive of single event upsets (SEUs) maintained by Inmarsat, one of the world's leading providers of global mobile satellite communications services. Inmarsat has operated its geostationary communication satellites and collected extensive satellite anomaly and telemetry data since 1990. Over the course of the past twenty years, the satellites have experienced more than 226 single event upsets (SEUs), a catch-all term for anomalies that occur in a satellite's electronics such as bit-flips, trips in power supplies, and memory changes in attitude control systems. While SEUs are seemingly random and difficult to predict, we correlate their occurrences to space weather phenomena, and specifically show correlations between SEUs and solar proton events (SPEs). SPEs are highly energetic protons that originate from solar coronal mass ejections (CMEs). It is thought that when these particles impact geostationary (GEO) satellites they can cause SEUs as well as solar array degradation. We calculate the associated statistical correlations that each SEU occurs within one day, one week, two weeks, and one month of SPEs at 10 – 10,000 MeV with energies greater than 10 particle flux units (pfu). However, we find that SPEs are most prevalent at solar maximum and that the SEUs on Inmarsat's satellites occur out of phase with the solar maximum. Ultimately, this suggests that SPEs are not the primary cause of the Inmarsat SEUs. A better understanding of the causal relationship between SPEs and SEUs will help the satellite communications industry develop component and operational space weather mitigation techniques as well as help the space weather community to refine radiation models.

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I. INTRODUCTION

The UK satellite telecommunications company, Inmarsat, currently operates ten satellites that provide global coverage for its mobile services. These satellites consist of three different generations. Inmarsat maintains an archive containing more than twenty years' worth of telemetry and satellite anomaly data from these three satellite generations.

The contents of Inmarsat's telemetry database include component health data such as total bus power and solar array current. The database also contains records of various anomalies including solid-state power amplifier (SSPA) failures, telemetry corruption, and single event upsets (SEUs). In this work, SEUs are categorized as seemingly random anomalies that occur in a satellite's electronics; for example: bit-flips, trips in power supplies, and memory changes in attitude control systems (ACS). For the two generations spanning from 1996 to 2012, there were 226 SEUs combined. The satellite fleets are referred to in this paper as A and B. Of these, fleet A experienced 28 SEUs, an average of 5.6 SEUs per satellite, and fleet B experienced 198, an average of 66 SEUs per satellite. SEUs are particularly interesting types of anomalies because they are hard to predict and, as our results show, dependent on hardware. As hardware technological capabilities grow and component form factors shrink, satellites are becoming increasingly more susceptible to radiation effects such as charging, radiation damage, and SEUs [1].

SEUs occur when highly energetic particles penetrate the surface and deposit charge into the electronics of a satellite [2]. Particles with energies greater than 10 MeV are capable of penetrating the surface of a satellite, and can produce damage equivalent to years of normal on-orbit operation [1]. In severe cases, SEUs can cause satellites to lose control and tumble, potentially leading to satellite failure. The two main sources of particles that cause SEUs are solar energetic protons and galactic cosmic rays (GCRs); the particles trapped within the magnetosphere do not possess sufficient energy to cause SEUs

for GEO communication satellites [3].

High-energy solar protons originate from coronal mass ejections (CMEs). The Sun emits CMEs at speeds of 1000 km/s. These particles take only 1.5 to 2 days to reach Earth [1] [4], and when the particles reach satellites at geostationary orbit they can cause SEUs and significantly contribute to solar array degradation [5].

High-energy protons can penetrate the Earth's magnetic field at the poles, crash into neutral atmospheric particles, and produce ion and electron pairs that temporarily increase the plasma density in the lowest regions of the ionosphere. This causes absorption of short-wave radio signals and widespread blackout of communications, sometimes called a polar cap absorption event. In addition, hazardous levels of high-energy particle radiation build up in the magnetosphere; this radiation can damage spacecraft microelectronics and pose a serious threat to the safety of astronauts. Energetic proton events can cause increased noise in photonics, total radiation dose problems, power panel damage, and single event upsets.

GCRs consist mostly of protons (84% hydrogen) along with alpha particles (15% Helium) and less than 1% of heavier nuclei [3] [1]. GCRs have energies up to 10^{14} MeV and occur out of phase with the eleven year solar cycle; the radiation from GCRs peaks at solar minimum and reaches a minimum at solar maximum [3] [6]. At solar minimum, the solar winds are low and allow GCRs to reach the magnetosphere. This does not usually occur at solar maximum because the solar winds inhibit the GCRs from entering a trajectory towards the magnetosphere and the geostationary satellites. We do not assess GCR data in this study, but plan address GCRs in future work, as strong GCRs can also cause significant SEUs [7].

While it is clear that both high-energy solar protons and GCRs are capable of causing SEUs, our analysis of Inmarsat's SEU data shows that sensitivity to SEUs is also a function of differences in hardware. Although the satellite fleets under consideration here have consistent designs within a fleet, it is important to also recognize that multiple copies of a single hardware device produced from a single manufacturer have been found to vary by as much as 15% [8].

To date, Inmarsat has not had any extended service interruptions, permanent equipment failures or suffered any reduction in satellite lifetime directly attributable to SEUs. While SEUs are challenging to predict, the Inmarsat satellites make use of automatic on-board error detection and correction functions, coupled with the use of automated ground system monitoring.

In Section 2 we analyze solar proton events and their occurrence rates throughout the 11-year solar cycle. In Section 3 we assess the SEUs on satellite fleet A and satellite fleet B, and statistically correlate them to the solar proton events

discussed in Section 2. We summarize our results and discuss our conclusions in Section 4.

II. SOLAR PROTON EVENTS

Large solar proton events (SPEs), while fairly uncommon, can cause severe radiation damage to spacecraft, because they excite protons and heavy ions to high energy levels capable of penetrating surrounding structures and shielding of satellite electronics. The energy levels required for this form of penetration are approximately 50–100 MeV/nucleon [8], however particles at 10 MeV can also contribute to surface charging and SEUs [1].

For this study, the solar proton event data comes from the NOAA Space Weather Prediction Center (<http://www.swpc.noaa.gov/ftpdir/indices/SPE.txt>). For each event, the proton fluxes are integrated five-minute averages recorded from Geostationary Operational Environmental Satellites (GOES), which have monitored and reliably collected space environment data since 1976. The start of a proton event is defined when three consecutive proton fluxes are recorded at a rate greater than or equal to ten particle flux units (pfu), where one pfu is a rate of one particle per square centimeter of detector area per steradian of solid angle scanned per second of time. The end of a proton event is the last instance the rate is above ten pfu. This definition accounts for multiple proton flares or interplanetary shock increases within one proton event.

In this section, we compare the occurrence of solar proton events to the solar cycle. Figure 1 shows the smoothed sunspot number and the 10 MeV–10,000 MeV solar proton events between 1996 and 2012. The sunspot number is a metric used to assess the overall strength and fluctuation of solar activity, such as solar flares and CMEs. The increase and decrease in sunspot number defines the solar maximum and solar minimum. At solar maximum there is an increased chance of solar flares, coronal mass ejections and other solar phenomena. However, even at solar minimum the Sun can produce damaging storms [9]. The solar magnetic activity cycle has a period of approximately eleven years.

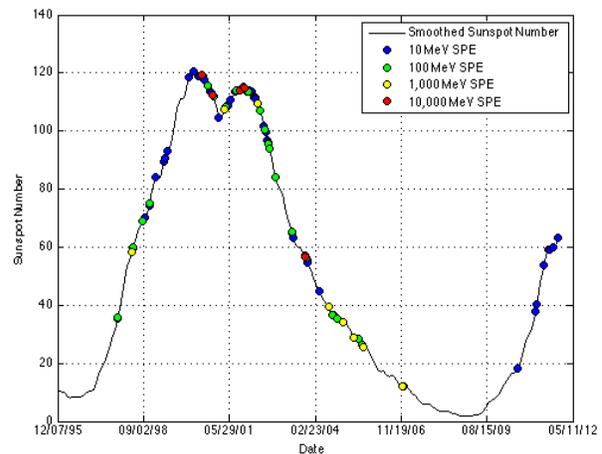


Figure 1: The solar cycle, the severity of SPEs, and the annual number of SPEs.

Solar proton events that involve 10 MeV particles with arrival rates greater than 10 pfu are considered significant events [10]. A list of solar proton events for particles between 10 MeV and 10,000 MeV and with rates greater than 10 pfu were recorded and can be found in Appendices A—D. The strongest and most frequent SPEs occur when the sunspot number is at a maximum; this is known as solar maximum, and occurs between 1998 and 2002.

In 2001 there were 22 SPEs and in 2002 there were 19 SPEs; these two years had the highest number of SPEs between 1996 and 2012. In 2001 and 2002, five of the six most severe SPEs, measuring 12,900 to 31,700 pfu, occurred during solar maximum. Significantly fewer SPEs occurred at solar minimum: from 2007 to 2009 there were zero SPEs.

For a seasonal perspective, the distribution of SPEs per month is shown in Figure 2. The highest number of SPEs occurred in November with sixteen SPEs. Interestingly for SEUs (instead of SPEs), the season around northern hemisphere winter solstice is the period of time with the fewest SEUs [5]. The months with the second highest number of SPEs were April and July, which both had twelve SPEs. For the northern hemisphere, April is just after vernal equinox, and July is just after summer solstice.

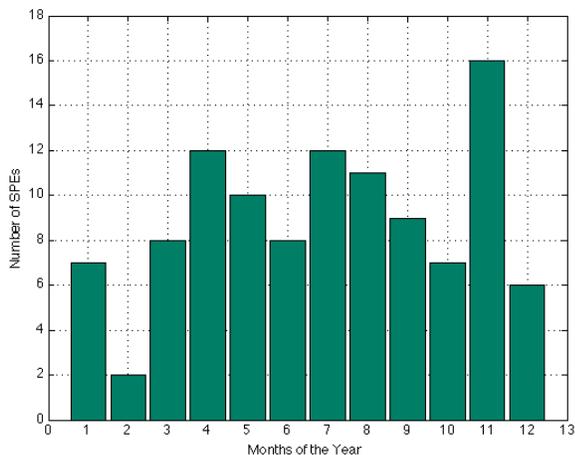


Figure 2: The number of SPEs per month measured from GOES, dates attached in Appendix A – Appendix D

Figure 3 shows the distribution of SPEs during the Bartels cycle. The Bartels cycle, similar to the Carrington cycle, is the 27-day cycle of the Sun’s rotation [11]. There is no obvious trend for this distribution. The day with the highest number of SPEs occurred on the third and 26th day of the Bartels cycle, and the day with the fewest number of SPEs occurs on the 22nd day of the cycle.

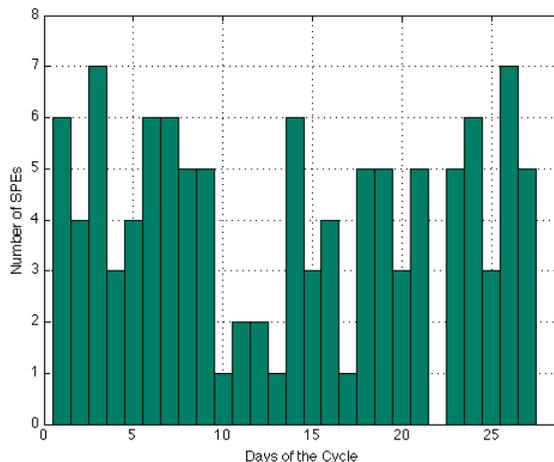


Figure 3: The number of SPEs per day of Bartels cycle measured from GOES, dates attached in Appendix A – Appendix D

III. SINGLE EVENT UPSETS

The causal relationship between high-energy protons and single event upsets has been studied since 1979, when two separate papers, Guenzler et al. and McNulty et al., presented evidence that high-energy protons were capable of causing SEUs [2]. A follow-up study analyzed the correlation between SEUs and the space environment that occurred on the NASA’s Tracking and Delay Relay Satellite System (TDRS-1) from 1984 to 1990 [3]. Similar to Inmarsat, TDRS-1 was designed to provide communications and high-data rate transmission services. Wilkinson [3] finds that protons with energies greater than 10 MeV—40 MeV tend to induce SEUs, and that SEUs are most likely to occur when protons with energies >50 MeV exceed a rate of 10 pfu [13]. However, Tylka et al. [8] states that in order to do a thorough correlation analysis of SEUs, both solar protons and heavy ions must be investigated.

The composition of the solar wind is a mixture of materials found in the solar plasma, ionized hydrogen (electrons and protons) with an 8% component of helium (alpha particles) and trace amounts of heavy ions and atomic nuclei: C, N, O, Ne, Mg, Si, S, and Fe that were ripped apart by heating of the Sun’s outer atmosphere, the corona [12]. This study only attempts to correlate solar protons to SEUs; analysis of alpha particles and heavy ions are planned in future work.

Figure 4 shows the distribution of SEUs per month. The highest number of SEUs occurs in April, followed by May. As shown in Figure 2, April was the month with the second highest number of SPEs, after November, which had the highest number of SPEs. For SEUs, November had the fourth lowest number of SEUs, followed by December, June, and February, which had the fewest number of SEUs.

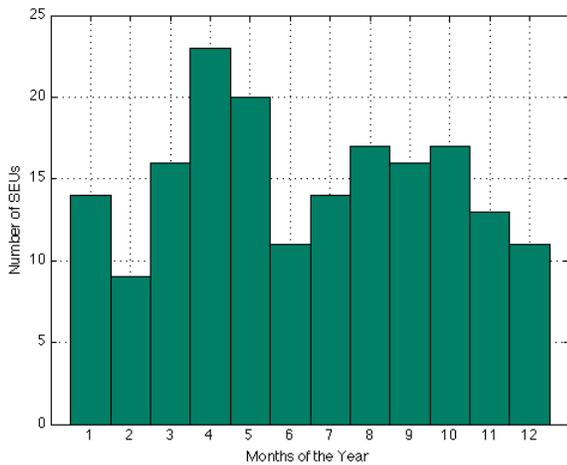


Figure 4: The number of SEUs per month measured from GOES, dates attached in Appendix A – Appendix D.

Choi et al. (2012) found that the majority of satellite anomalies occurred in spring (March-May) and fall (September – October). While we cannot state that all types of anomalies follow this trend, the occurrence of the majority of SEUs on Inmarsat’s satellite fleets are consistent with the results in Choi et al., and occur in the spring and fall [13].

Figure 5 shows the distribution of SEUs during the Bartels 27-day solar cycle. Similar to Figure 3, there does not seem to be any clear and consistent trends for SEUs during the Bartels solar cycle. The day with the highest number of SEUs occurs on the 21st day of the cycle, and the day with the fewest number of SEUs occurs on the 20th day of the cycle.

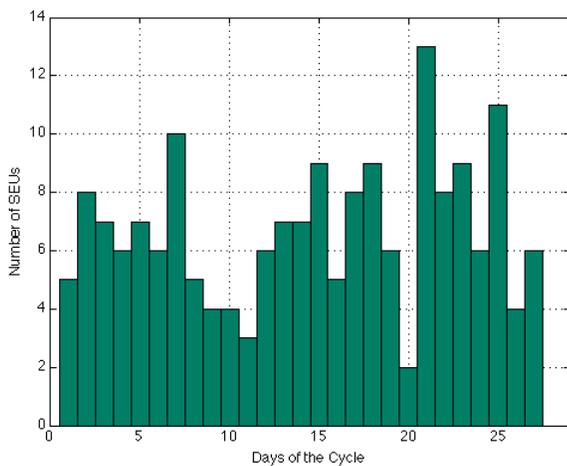


Figure 5: The number of SEUs per day of Bartels cycle measured from GOES, dates attached in Appendix A – Appendix D

SEUs can occur at any time during the solar cycle, but tend to occur nearer to solar minimum, as shown in Figure 6, which is consistent with previous investigations [14] [15].

As discussed in Section II, Figure 1 shows that more solar proton events occur at solar maximum than at solar minimum. Figure 6 shows the SEUs that occurred on Fleet A between 1996 and 2012; the different colors represent the five different satellites in fleet A. The SEUs tend to occur most often at solar minimum. Specifically, the largest number of SEUs per year occurred in 1997, 2005 and 2009. However, between 1998 and 2000, the years leading up to solar maximum, there were also high levels of SEUs, and there were no SEUs in both 2001 near solar maximum as well as in 2008, near solar minimum.

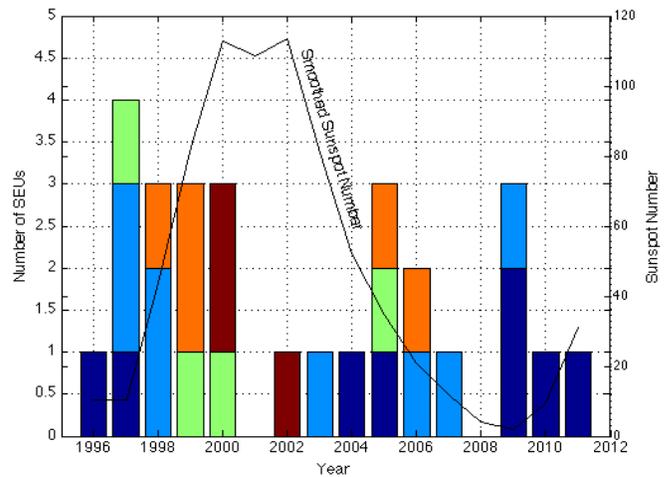


Figure 6: The solar cycle and the annual number of SEUs per satellite for satellite fleet A. The different colors represent the five different satellites in fleet A.

One possible explanation why there are zero SEUs at the 2008 solar minimum is the fact that there are only a few SEU events for satellite fleet A. Satellite fleet A only experienced 28 total SEUs, in comparison with fleet B, which experienced 198 SEUs. Figure 7 is a scatter plot of the twenty-eight SEUs that occurred on fleet A. The SEUs are shown as a black asterisk for solar proton events ranging from 10–10,000 MeV and rates greater than 10 pfu. There were zero SPEs between 2006 and 2010, which is approximately solar minimum.

As shown in Figure 1, a high concentration of SPEs occurs during solar maximum, or between 1998 and 2002. During this time, ten of the twenty-eight SEUs on fleet A occur. The other eighteen SEUs occur during the solar minimum of Cycle 23 and Cycle 24.

Table 1 shows the SEUs on Satellite fleet A that occur one day, one week, two weeks, and one month before solar proton events ranging from 10 MeV to 10,000 MeV with rates greater than 10 pfu.

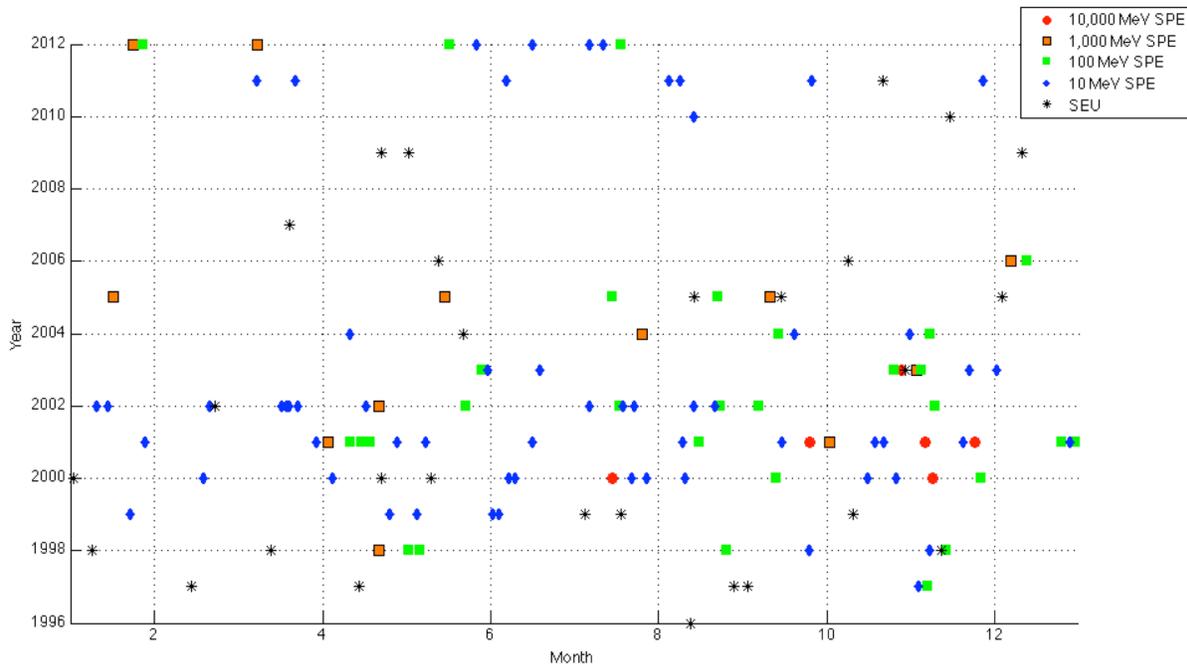


Figure 7: Fleet A SEUs plotted with 10—10,000 MeV SPEs that have rates greater than 10 pfu from 1996—2012. The red circles are 10,000 MeV SPEs, the orange squares are 1,000 MeV SPEs, the green squares are 100 MeV SPEs, the blue diamonds are 10 MeV SPEs and the black asterisks are SEUs. Eighteen of the twenty-eight SEUs on Fleet A occur at period of solar minimum, ten SEUs occur at solar maximum.

Table 1: Fleet A SEUs that occur 1 day, 1 week, 2 weeks, and 1 month before solar proton events of 10 MeV – 10,000 MeV with rates greater than 10 pfu

Solar Proton Events	The number of SEUs that occur before an SPE on fleet A			
	< 1 day before	< 1 week before	< 2 weeks before	< 1 month before
10 MeV	0	0	0	6
100 MeV	0	2	3	3
1,000 MeV	0	1	1	2
10,000 MeV	0	0	0	0

Fleet A encountered 28 SEUs, and eleven of these occurred up to one month before an SPE. Of these eleven, four SEUs occurred two weeks before an SPE and three occurred one week before an SPE. No events occurred one day before an SPE. Should a single SEU occur within one month of both a 10 MeV SPE and within one month of a later 100 MeV SPE it is counted twice, once for each of the different energy levels.

Table 2 shows the number of SEUs on fleet A that occur *after* an SPE for the same periods of time in Table 1.

Table 2: Fleet A SEUs that occur 1 day, 1 week, 2 weeks, and 1 month after solar proton events of 10 MeV – 10,000 MeV with rates greater than 10 pfu

Solar Proton Events	SEU occurs after the following periods of SPE on fleet A			
	< 1 day after	< 1 week after	< 2 weeks after	< 1 month after
10 MeV	0	2	2	5
100 MeV	0	1	1	2
1,000 MeV	0	1	1	1
10,000 MeV	1	1	1	1

Of the 28 SEUs encountered by fleet A, nine SEUs occurred one month after a solar proton event. Out of these nine SEUs, five occurred two weeks and five occurred one week after an SPE, and one SEU occurred one day after an SPE.

The number of SEUs that occurred within the specified periods of an SPE for does not necessarily suggest that a significant relationship exists between the occurrence of an SEU and SPEs. While coronal mass ejections take approximately 1.5—2 days to reach Earth, the amount of time between interactions of energetic particles with components that may eventually lead to a SEU is not well understood. The data shown here look at these statistics to get a sense for the general occurrence rates.

Figure 8 shows the smoothed sunspot number and the annual number of SEUs for satellite fleet B. Fleet B experienced 198 SEUs between 1996 and 2012. The first SEU occurred in 2005, at a period near solar maximum.

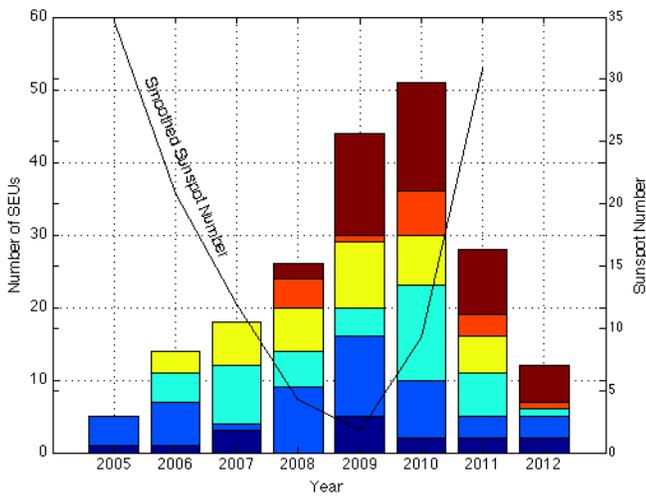


Figure 8: The solar cycle and the annual number of SEUs per satellite for satellite fleet B. The six different colors represent the local and remote computers on each of the three satellites in fleet B. An inverse correlation exists between the solar cycle and SEUs.

For fleet B, it is clear that more SEUs occur during solar minimum, compared with solar maximum. This inverse relationship strongly suggests that the single event upsets were not primarily caused by solar energetic protons, as we have shown in Figure 1 that more solar proton events occur at solar maximum. Thus, a more likely source of these SEUs could be GCRs, which primarily occur at solar minimum [1].

Similar to Figure 7, Figure 9 is a scatter plot of the SEUs, shown with a black asterisk, and the 10 MeV—100 MeV SPEs with rates greater than 10 pfu for fleet B.

The highest concentration of SEUs for fleet B occurs between 2006 and 2012, and coincides with more than four years of zero severe SPE activity. The years 2006 to 2012 also span across solar minimum.

Table 3 shows the SEUs on Satellite fleet B that occur one day, one week, two weeks, one month before solar proton events ranging from 10 MeV to 10,000 MeV with energy greater than 10 pfu.

Table 3: Fleet B SEUs that occur 1 day, 1 week, 2 weeks, and 1 month before solar proton events of 10 MeV – 10,000 MeV with rates greater than 10 pfu

Solar Proton Events	SEU occurs before the following periods of SPE on fleet B			
	< 1 day before	< 1 week before	< 2 weeks before	< 1 month before
10 MeV	0	4	7	19
100 MeV	0	1	3	6
1,000 MeV	0	1	2	8
10,000 MeV	0	0	0	0

Of the 198 SEUs that occurred on fleet B, 33 occurred one month before an SPE. Twelve of the 33 SEUs occurred two weeks before an SPE, and six occurred one week before an SPE. Zero SEUs occurred one day before an SPE.

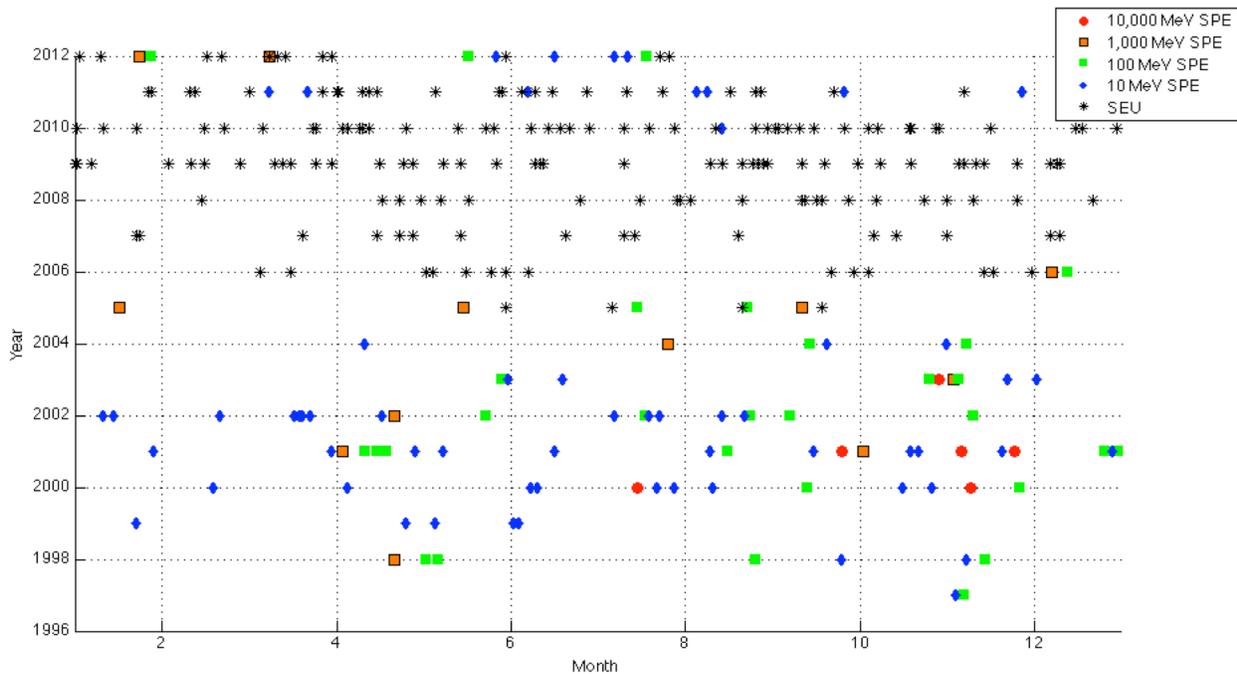


Figure 9: Fleet B SEUs plotted with 10—10,000 MeV SPEs that have rates greater than 10 pfu from 1996—2012. The red circles are 10,000 MeV SPEs, the orange squares are 1,000 MeV SPEs, the green squares are 100 MeV SPEs, the blue diamonds are 10 MeV SPEs and the black asterisks are SEUs. A clear inverse correlation exists between the occurrence of SEUs and SPEs, as only 33 of the 198 SEUs on Fleet B occurred one month before an SPE.

Table 4 shows the number of SEUs on fleet B that occur *after* the same periods of time in Table 3.

Table 4: Fleet B SEUs that occur 1 day, 1 week, 2 weeks, and 1 month after solar proton events of 10 MeV – 10,000 MeV with rates greater than 10 pfu

Solar Proton Events	SEU occurs after the following periods of SPE on fleet B			
	1 day after	1 week after	2 weeks after	1 month after
10 MeV	0	3	13	37
100 MeV	1	2	4	9
1,000 MeV	1	6	6	12
10,000 MeV	0	0	0	0

Out of the total 198 SEUs, fifty-four SEUs occurred within one month of an SPE. Twenty-three of the fifty-four occur two weeks after an SPE, eleven SEUs occurred one week after an SPE, and two SEUs occurred one day after an SPE.

Figure 10 shows the age of the satellite at the time of the SEU from 1996 to 2012. There are eight total satellites but the eleven different colors portray the five satellites of fleet A and separately track the local and remote computers on the three satellites of fleet B. Throughout this time the total number of satellites increases from one to eight and therefore the age of the satellites in 2012 ranges from five to fifteen years old. Aside from an overall bifurcation between fleet A and fleet B, and the influence of the 11-year solar cycle, there does not appear to be an obvious correlation between SEUs and the age of the satellite. It has been suggested that satellite anomalies are more prevalent in the first two years of operation due to “burn-in” and transients from the launch and orbital positioning activities. However, based on Figure 10, this does not appear to be a substantial contributor to the SEUs on these two satellite fleets.

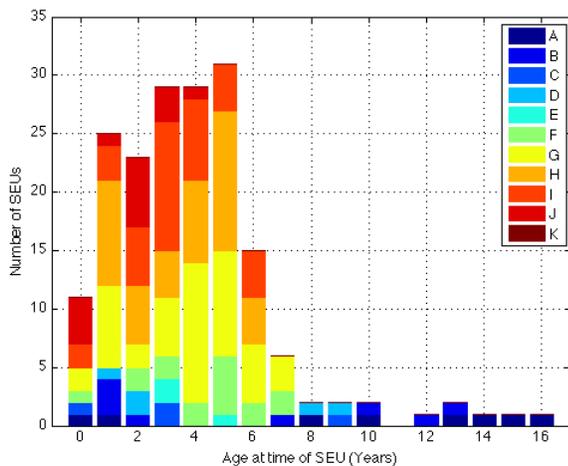


Figure 10: The age (years) of the five satellites in fleet A and the three satellites in fleet B at the time of the SEU. For fleet B the SEUs on the local and remote computers are plotted separately, which creates 11 separate categories,

rather than eight.

IV. CONCLUSIONS

Inmarsat has maintained nominal system operations through more than 226 single event upsets on two of its satellite fleets since 1996. Satellite fleet A experienced 28 SEUs, and satellite fleet B experienced 198 SEUs. The occurrence of these SEUs was compared to the occurrence of solar proton events. Ten MeV SPEs with rates greater than 10 pfu are considered significant and this level was used as the reference for defining an event in this study.

At solar maximum we show there are fewer SEUs than at solar minimum, which is when the largest number of SEUs occur. We also show that the largest number of SPEs occurred at solar maximum, and fewest at solar minimum. Therefore, SEUs on the Inmarsat fleet were not found to correlate to solar proton events.

There is not a clear correlation between SPEs or SEUs and the Bartels solar cycle or the month in which either of these phenomena occur.

For fleet A, eleven of the total twenty-eight SEUs occurred one month *before* an SPE. Of these eleven, four SEUs occurred two weeks before an SPE and three occurred one week before an SPE. No events occurred one day before an SPE. Nine SEUs occurred one month *after* an SPE. Five of these occurred two weeks after an SPE, five occurred one week after an SPE, and one occurred one day after an SPE. Although fleet A only experienced a total of 28 SEUs or 5.6 SEUs per satellite, from fleet A we conclude that SEUs do not appear to have a direct correlation with energetic proton events.

Adding to our analysis the data from fleet B, which had 198 SEUs, an average of 66 per satellite, we find a clear inverse relationship between SEUs and SPEs. This finding suggests that solar energetic protons did not primarily cause the SEUs. Thirty-three of the 198 SEUs occurred one month before an SPE. Twelve of the 33 SEUs occurred two weeks before an SPE, and six occurred one week before an SPE. Zero SEUs occurred one day before an SPE. Fifty-four SEUs occurred within one month *after* an SPE. Twenty-three of these occurred two weeks after an SPE, eleven occurred one week after an SPE, and two SEUs occurred one day after an SPE.

We also note that there is not a clear correlation between SEUs and the age of the satellite.

The next step in this work is to investigate the correlation between SEUs and GCRs as a function of the 11-year solar cycle, as well as to consider correlations between SEUs and heavier elements. Further, the results appear to show a notable difference in the susceptibility of fleet A and fleet B to SEUs. Closer analysis and attention paid to the particular subsystems and components most susceptible to SEUs would be valuable

in explaining this difference and providing information about what mitigation methods can be employed, and information about which materials are susceptible to which types and energies of particles.

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BIOGRAPHIES



Whitney Q. Lohmeyer received a B.S. in Aerospace Engineering from North Carolina State University in 2011, and will obtain a Masters Candidate in Aerospace Engineering at Massachusetts Institute of Technology in February 2013. After obtaining her Masters she will pursue a PhD at MIT in Aerospace Engineering, researching the effect of the space environment on geostationary satellites.



Kerri Cahoy received a B.S. in Electrical Engineering from Cornell University in 2000, an M.S. in Electrical Engineering from Stanford University in 2002, and a Ph.D. in Electrical Engineering from Stanford University in 2008. After working as a Senior Payload and Communication Sciences Engineer at Space Systems Loral, she completed a NASA Postdoctoral Program

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APPENDIX

Appendix A: SPEs with Proton Flux > 10,000 MeV

Date	Year	Proton Flux
15-Jul	2000	24000
9-Nov	2000	14800
25-Sep	2001	12900
6-Nov	2001	31700
24-Nov	2001	18900
29-Oct	2003	29500

Appendix B: SPEs with Proton Flux > 1,000 MeV

Date	Year	Proton Flux
21-Apr	1998	1700
3-Apr	2001	1110
2-Oct	2001	2360
21-Apr	2002	2520
3-Nov	2003	1570
26-Jul	2004	2086
17-Jan	2005	5040
15-May	2005	3140
11-Sep	2005	1880
7-Dec	2006	1980
24-Jan	2012	6310
8-Mar	2012	6530

Appendix C: SPEs with Proton Flux > 100 MeV

Date	Year	Proton Flux
7-Nov	1997	490
2-May	1998	150
6-May	1998	210
26-Aug	1998	670
14-Nov	1998	310
13-Sep	2000	320
26-Nov	2000	940
11-Apr	2001	355
15-Apr	2001	951
18-Apr	2001	321
16-Aug	2001	493
26-Dec	2001	779
31-Dec	2001	108
23-May	2002	820
17-Jul	2002	234
24-Aug	2002	317
7-Sep	2002	208
10-Nov	2002	404
29-May	2003	121
26-Oct	2003	466
5-Nov	2003	353
14-Sep	2004	273
8-Nov	2004	495
15-Jul	2005	134
23-Aug	2005	330
13-Dec	2006	698
28-Jan	2012	796
17-May	2012	255
18-Jul	2012	136

Appendix D: SPEs with Proton Flux > 10 MeV

Date	Year	Proton Flux
4-Nov	1997	72
25-Sep	1998	44
8-Nov	1998	11
23-Jan	1999	14
25-Apr	1999	32
5-May	1999	14
2-Jun	1999	48
4-Jun	1999	64
18-Feb	2000	13
5-Apr	2000	55
8-Jun	2000	84
10-Jun	2000	46
22-Jul	2000	17
28-Jul	2000	18
11-Aug	2000	17
16-Oct	2000	15
26-Oct	2000	15
29-Jan	2001	49
30-Mar	2001	35
28-Apr	2001	57
8-May	2001	30
16-Jun	2001	26
10-Aug	2001	17
15-Sep	2001	11
19-Oct	2001	11
22-Oct	2001	24
20-Nov	2001	34
29-Dec	2001	76
11-Jan	2002	91
15-Jan	2002	15
20-Feb	2002	13
17-Mar	2002	13
19-Mar	2002	53
20-Mar	2002	19
23-Mar	2002	16
17-Apr	2002	24
7-Jul	2002	22
19-Jul	2002	13
23-Jul	2002	28
14-Aug	2002	26
22-Aug	2002	36
31-May	2003	27

19-Jun	2003	24
22-Nov	2003	13
2-Dec	2003	86
11-Apr	2004	35
20-Sep	2004	57
1-Nov	2004	63
14-Aug	2010	14
8-Mar	2011	50
22-Mar	2011	14
7-Jun	2011	72
5-Aug	2011	96
9-Aug	2011	26
26-Sep	2011	35
27-Nov	2011	80
27-May	2012	14
16-Jun	2012	14
7-Jul	2012	25
12-Jul	2012	96